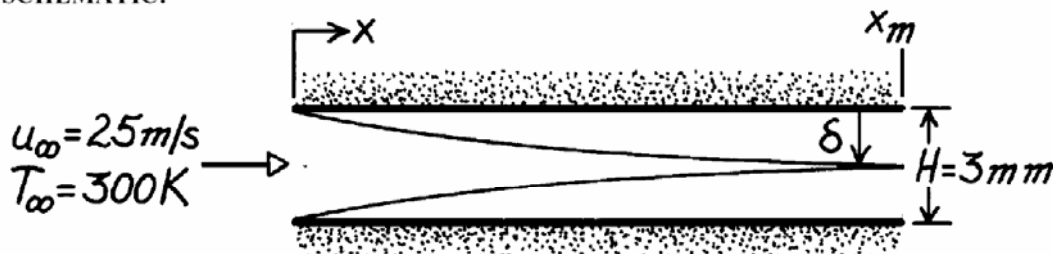


PROBLEM 7.3

KNOWN: Velocity and temperature of air in parallel flow over a flat plate.

FIND: (a) Velocity boundary layer thickness at selected stations. Distance at which boundary layers merge for plates separated by $H = 3 \text{ mm}$. (b) Surface shear stress and $v(\delta)$ at selected stations.

SCHEMATIC:



ASSUMPTIONS: (1) Steady flow, (2) Boundary layer approximations are valid, (3) Flow is laminar.

PROPERTIES: Table A-4, Air (300 K, 1 atm): $\rho = 1.161 \text{ kg/m}^3$, $\nu = 15.89 \times 10^{-6} \text{ m}^2/\text{s}$.

ANALYSIS: (a) For laminar flow,

$$\delta = \frac{5x}{\text{Re}_x^{1/2}} = \frac{5}{(u_\infty/\nu)^{1/2}} x^{1/2} = \frac{5x^{1/2}}{(25 \text{ m/s}/15.89 \times 10^{-6} \text{ m}^2/\text{s})^{1/2}} = 3.99 \times 10^{-3} x^{1/2}.$$

$x \text{ (m)}$	0.001	0.01	0.1
$\delta \text{ (mm)}$	0.126	0.399	1.262

Boundary layer merger occurs at $x = x_m$ when $\delta = 1.5 \text{ mm}$. Hence

$$x_m^{1/2} = \frac{0.0015 \text{ m}}{3.99 \times 10^{-3} \text{ m}^{1/2}} = 0.376 \text{ m}^{1/2} \quad x_m = 141 \text{ mm.} \quad <$$

(b) The shear stress is

$$\tau_{s,x} = 0.664 \frac{\rho u_\infty^2 / 2}{\text{Re}_x^{1/2}} = 0.664 \frac{\rho u_\infty^2 / 2}{(u_\infty/\nu)^{1/2} x^{1/2}} = \frac{0.664 \times 1.161 \text{ kg/m}^3 (25 \text{ m/s})^2 / 2}{(25 \text{ m/s}/15.89 \times 10^{-6} \text{ m}^2/\text{s})^{1/2} x^{1/2}} = \frac{0.192}{x^{1/2}} \left(\text{N/m}^2 \right).$$

$x \text{ (m)}$	0.001	0.01	0.1
$\tau_{s,x} \text{ (N/m}^2\text{)}$	6.07	1.92	0.61

The velocity distribution in the boundary layer is $v = (1/2) (\nu u_\infty / x)^{1/2} (\eta df/d\eta - f)$. At $y = \delta$, $\eta \approx 5.0$, $f \approx 3.24$, $df/d\eta \approx 0.991$.

$$v = \frac{0.5}{x^{1/2}} \left(15.89 \times 10^{-6} \text{ m}^2/\text{s} \times 25 \text{ m/s} \right)^{1/2} (5.0 \times 0.991 - 3.28) = (0.0167 / x^{1/2}) \text{ m/s.}$$

$x \text{ (m)}$	0.001	0.01	0.1
$v \text{ (m/s)}$	0.528	0.167	0.053

COMMENTS: (1) $v \ll u_\infty$ and $\delta \ll x$ are consistent with BL approximations. Note, $v \rightarrow \infty$ as $x \rightarrow 0$ and approximations breakdown very close to the leading edge. (2) Since $\text{Re}_{x_m} = 2.22 \times 10^5$, laminar BL model is valid. (3) Above expressions are approximations for flow between parallel plates, since $du_\infty/dx > 0$ and $dp/dx < 0$.